



# Ares I-X Overview – The First Chapter in the Next Great Adventure

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# Agenda

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  - Upper Stage Simulator (USS)
  - Roll Control System (RoCS)
  - First Stage
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- ◆ **Vehicle Assembly Building (VAB) Operations**
- ◆ **Processing Flow**
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- ◆ **Summary**





# Ares Launch Vehicles Background



- ◆ **Deliver crew and cargo for missions to International Space Station (ISS) and to Moon and beyond**
- ◆ **Continuing progress toward design, component testing, and early flight testing**
- ◆ **Ares I Crew Launch Vehicle**
  - Will carry 6 crew to ISS, 4 to Moon
  - First flight test 2009
  - Initial Operating Capability 2015
- ◆ **Ares V Cargo Launch Vehicle**
  - Will launch Earth departure stage and Altair lunar lander to low Earth orbit for lunar missions
  - Largest launch vehicle ever designed
  - Will begin detailed development work in 2011

# Ares I-X Background

- ◆ **Ares I-X is a development test flight to provide engineering data to inform the design of the Ares I prior to CDR**

*Ares I will replace the Space Shuttle which is scheduled for 2010 retirement*

- ◆ **Ares I-X is an uncrewed, sub-orbital development flight test**
- ◆ **Ares I-X will provide opportunity to test ground facilities and operations at NASA's Kennedy Space Center**





# Ares I-X Flight Test Objectives



**P(1) Demonstrate control** of a dynamically similar, integrated Ares I/Orion, using Ares I relevant ascent control algorithms

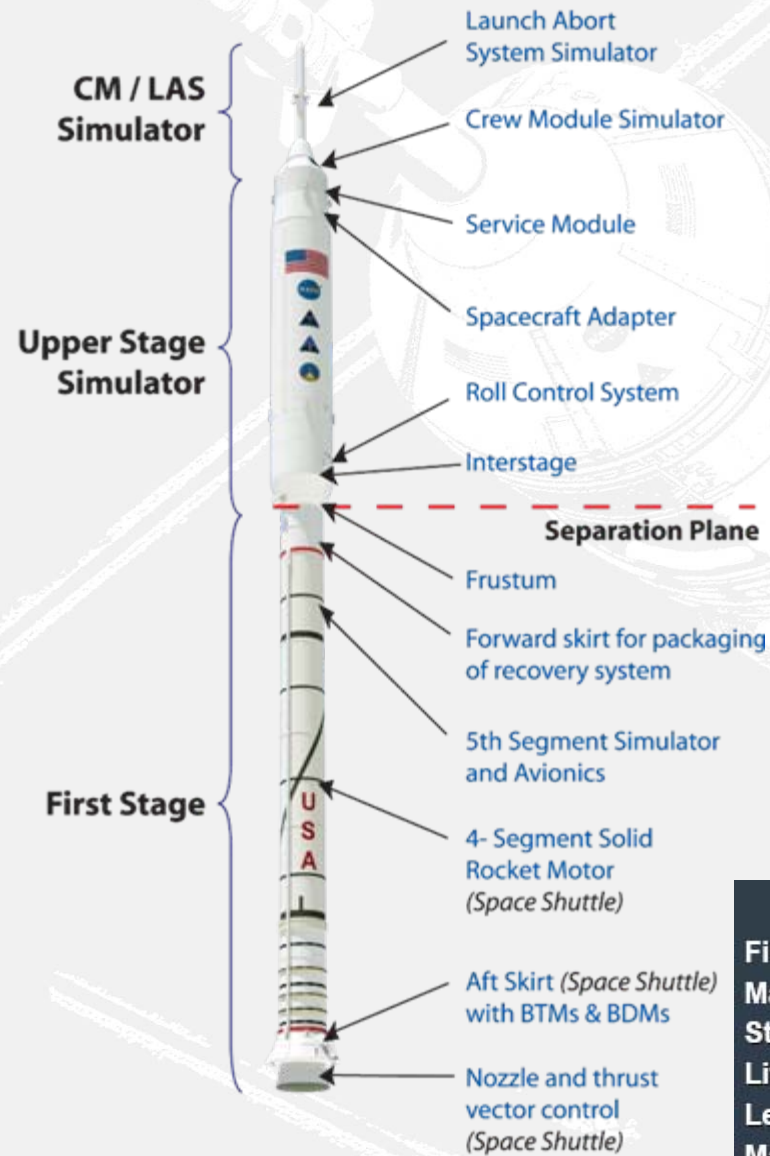
**P(2) Perform an in-flight separation/staging** event between a Ares I-similar First Stage and a representative Upper Stage

**P(3) Demonstrate assembly and recovery** of a new Ares I-like First Stage element at KSC

**P(4) Demonstrate First Stage separation sequencing, and quantify First Stage atmospheric entry dynamics, and parachute performance**

**P(5) Characterize magnitude of integrated vehicle roll torque** throughout First Stage flight

# Vehicle Overview



## ◆ Combines proven space flight and simulated hardware

### • Space flight hardware includes:

- Four-segment solid rocket booster (Space Shuttle)
- Atlas V-based avionics
- Roll control system (Peacekeeper)
- Separation system (Space Shuttle)
- Parachutes deceleration system (Space Shuttle)
- Booster deceleration and tumble motors (Space Shuttle)
- Developmental flight instrumentation

### • Simulator hardware

- Upper stage
- Orion crew module
- Launch abort system
- Fifth segment of booster

	Ares I-X	Ares I
<b>First Stage Max. Thrust (vacuum):</b>	14.1M N (3.13M lbf)	15.8M N (3.5M lbf)
<b>Max. Speed:</b>	Mach 4.7	Mach 5.84
<b>Staging Altitude:</b>	39,624 m (130,000 ft)	57,453 m (188,493 ft)
<b>Liftoff Weight:</b>	834k kg (1.8M lbm)	927k kg (2.0M lbm)
<b>Length:</b>	99.1 m (327 ft)	99 m (325 ft)
<b>Max. Acceleration:</b>	2.46 g	3.79 g



# Ares I-X Development Flight Test

**P2) Perform in-flight separation/staging event at 124 sec ~ 130,000 feet**

~ 150,000 feet

Vehicle Height:	327 feet
Weight at Ignition:	1.8 M-lbm
Max. Acceleration:	2.5 g's
Max. Speed:	Mach 4.7

**P4) Demonstrate FS entry dynamics and sequencing of events (parachute deployment, etc.)**

**P5) Characterize integrated vehicle roll torque**

in-flight separation plane

**P1) Demonstrate controllability**

**P3) Demonstrate assembly and recovery of an Ares I similar FS**

**USS/CM/LAS  
Uncontrolled descent and impact**

**Booster, parachutes and recovery**





# Orion Crew Module/ Launch Abort System (CM/LAS) Simulator



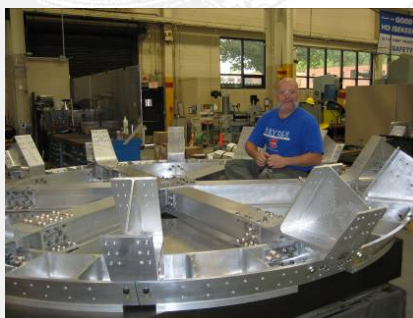
**LAS Party Hat Assembly**



**LAS Tube Machining**



**CM Lower Ring**



**CM Upper Ring**



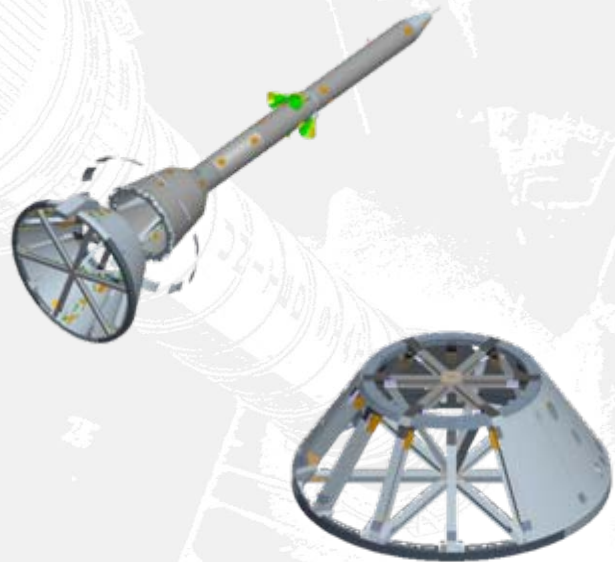
**CM/LAS Transport Vehicle**



**LAS Transition Fitting**



- ◆ Outer mold line (OML) resembles earlier Ares I design due to flight test schedule
- ◆ Developmental flight instrumentation sensors will measure aerodynamic and acoustic loads
- ◆ Developed at the NASA Langley Research Center, Hampton, VA





# Upper Stage Simulator (USS)

- ◆ **USS is a mass and Outer Mold Line (OML) simulator**
- ◆ **Hardware includes:**
  - Interstage (IS) Simulator
  - Upper Stage (US) Simulator
  - Spacecraft Adapter (SA) Simulator
  - Service Module (SM) Simulator
- ◆ **Developed at the NASA Glenn Research Center, Cleveland, OH**



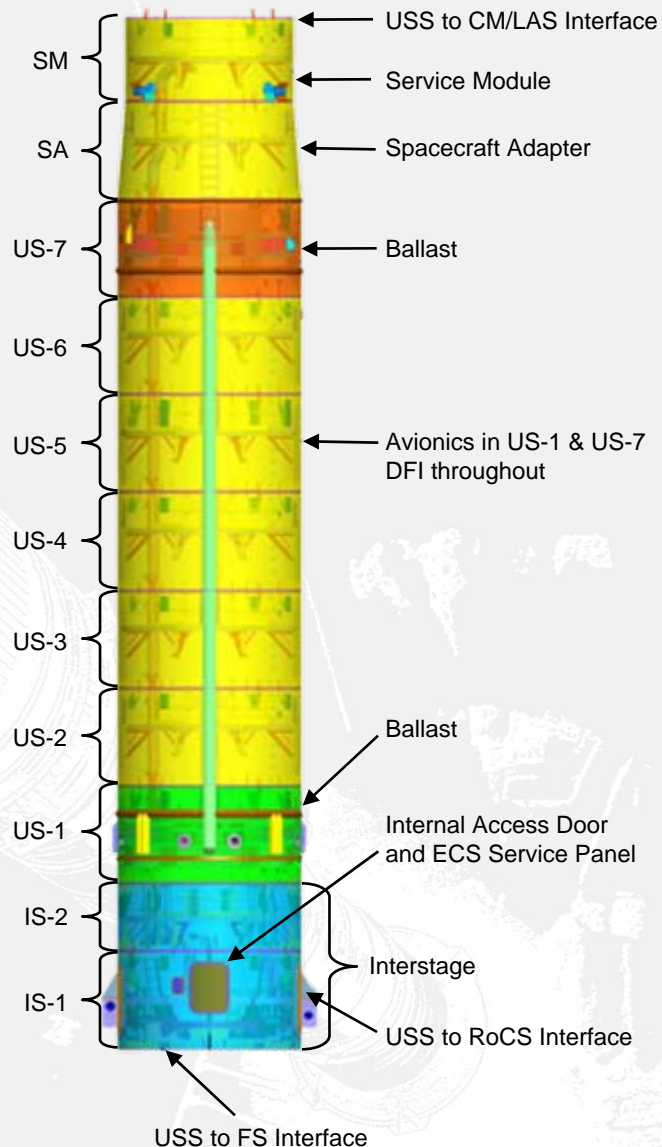
**Flange Machining**



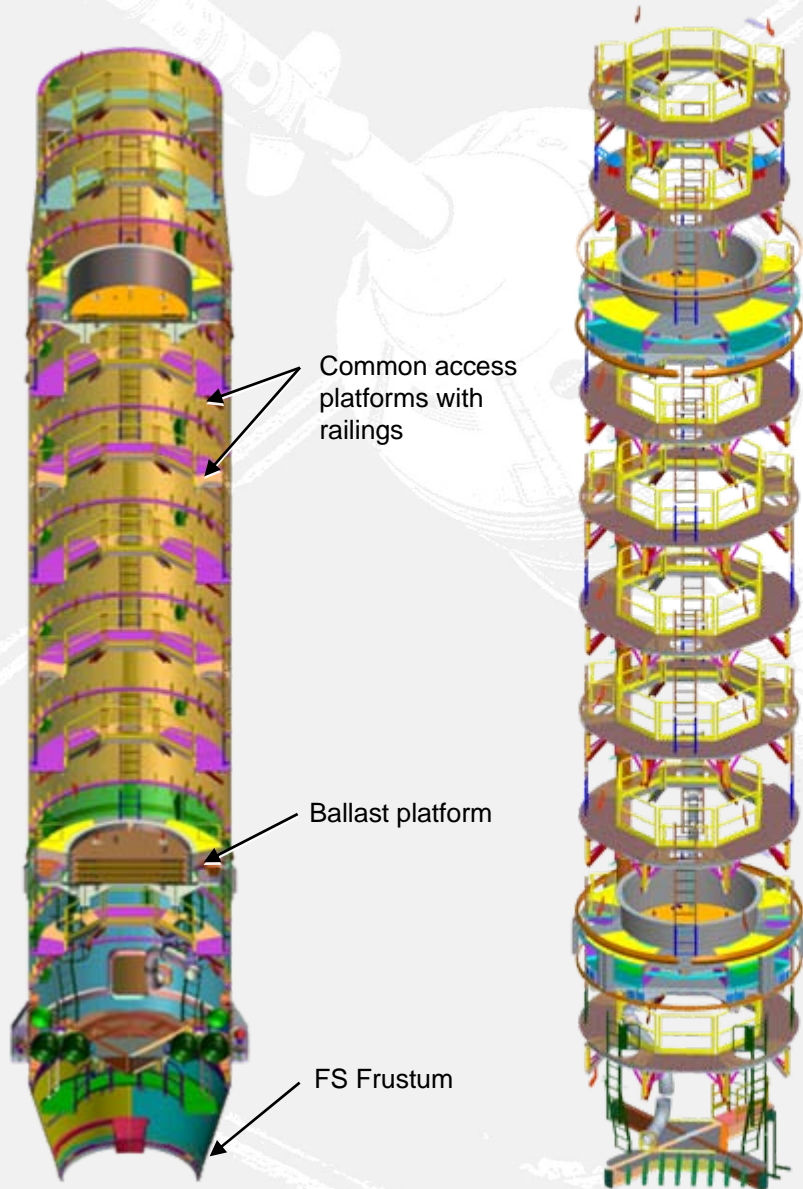
**Support Welding**



**Super Stack**



# USS Internal Access Concept

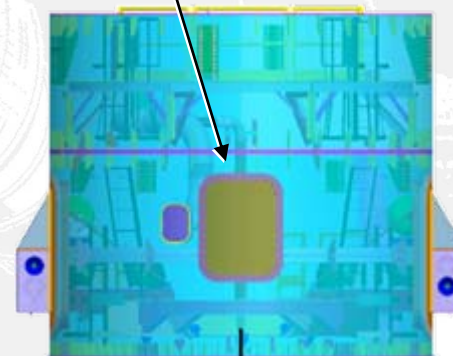


◆ Provides access from the Frustum to the CM/LAS

◆ Door in the IS-1 segment

- Internal access platforms and ladders
- Provides Environmental Control System (ECS) ductwork to maintain a safe work temp, air flow and controlled humidity

Internal access door



IS-1 Segment



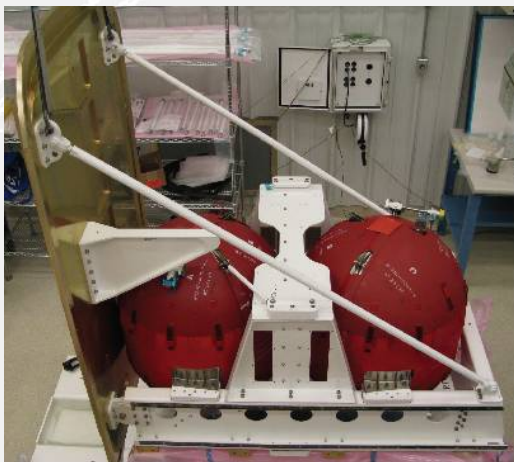
# Roll Control System (RoCS)



**Propellant Tank**



**Pressurization System**



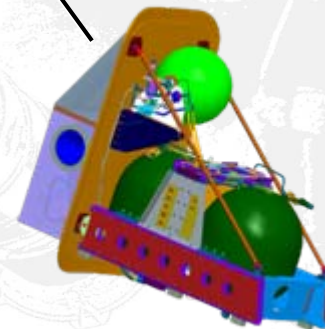
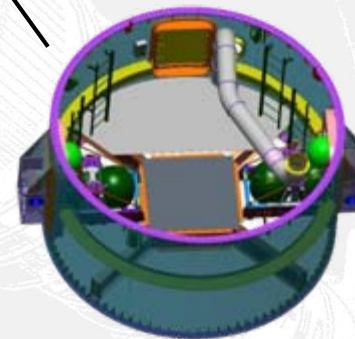
**Cold Flow Assembly**



**Thruster**



- ◆ Provides post-launch 90-degree roll and mitigation against adverse roll torques
- ◆ Modular propulsion system housed in the Ares I-X USS Interstage
- ◆ Proven space hardware harvested from Peacekeeper 4th Stage
- ◆ Managed at the NASA Marshall Space Flight Center, Huntsville, AL



# First Stage



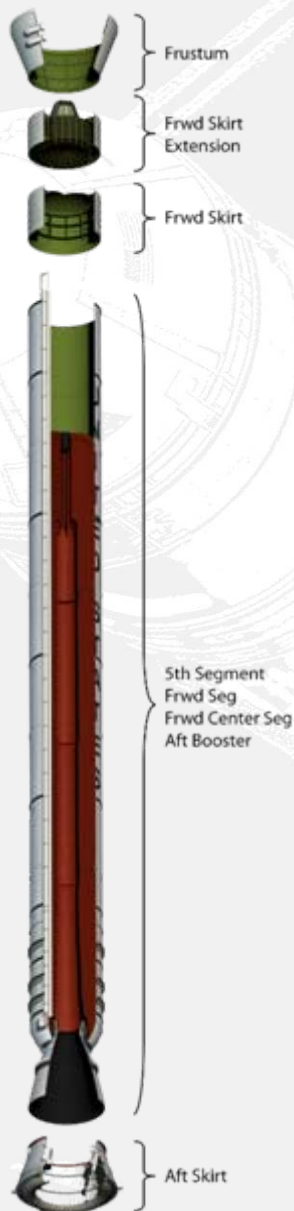
**Main Parachute**



**Frustum Forward Ring**



**Aft Skirt at ARF**



## ◆ Heritage Hardware

- 4 Segment Reusable Solid Rocket Motor (RSRM) w/Nozzle
- Thrust Vector Control (TVC)
- Flight Termination System (FTS)
- Nose Cap w/Thrusters
- Booster Separation Motors (BSMs)

## ◆ Modified Heritage Hardware

- Shuttle Derived Avionics
- Aft Skirt

## ◆ New Developments for Ares I-X

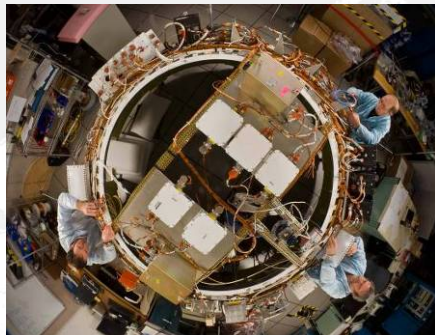
- Fifth Segment Simulator (5SS)
- Forward Skirt (FS)
- Forward Skirt Extension (FSE)
- Main Parachute Support Structure (MPSS)
- Frustum

## ◆ Ares I Designs

- Parachutes
- FTS Extension to Aft Segment

## ◆ Managed at the NASA Marshall Space Flight Center, Huntsville, AL



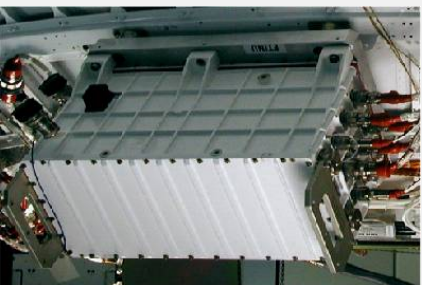


**Avionics SIL**



**FSAM Assembly**

**ATVC Testing**



**FTINU (Flight Computer)**

**FSAM Harness Template**



♦ **Primary avionics subsystems:**

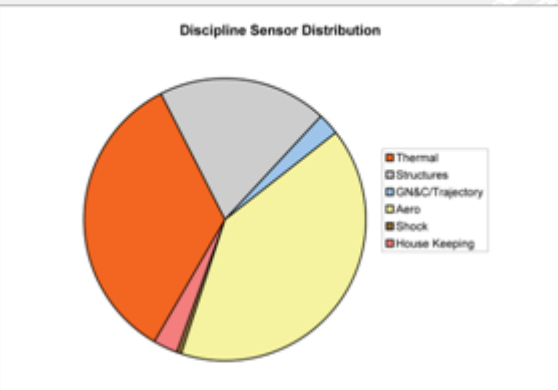
- FSAM (located in First Stage fifth segment)
- Guidance & Control System
- Ground Command, Control, and Communication (GC3)

♦ **Managed at the NASA Marshall Space Flight Center, Huntsville, AL**

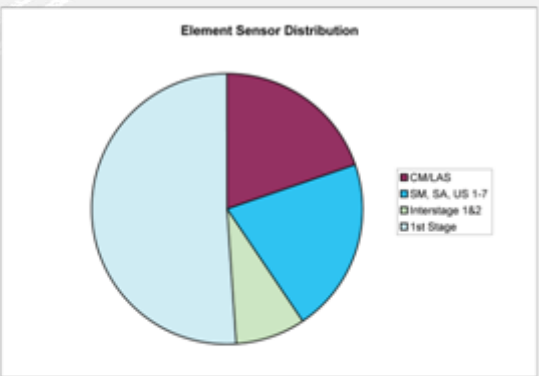


**FSAM**

## Sensor Summary Appendix

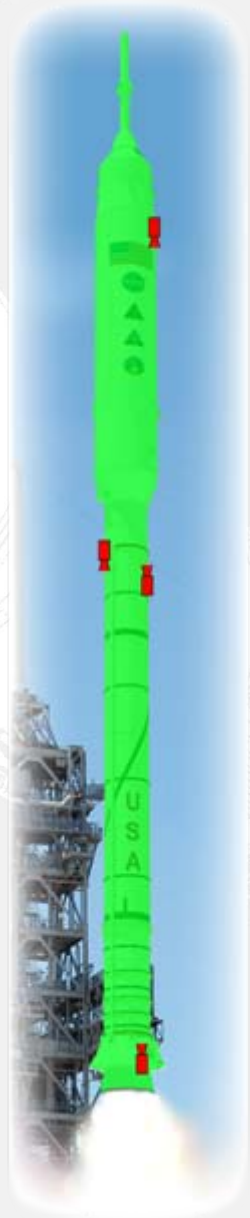


Discipline	Number of Sensors	% of Total
Thermal	256	34%
Structures	147	20%
GN&C/Trajectory	18	2%
Aero	305	41%
Shock	4	1%
House Keeping	21	3%
Total	751	100%

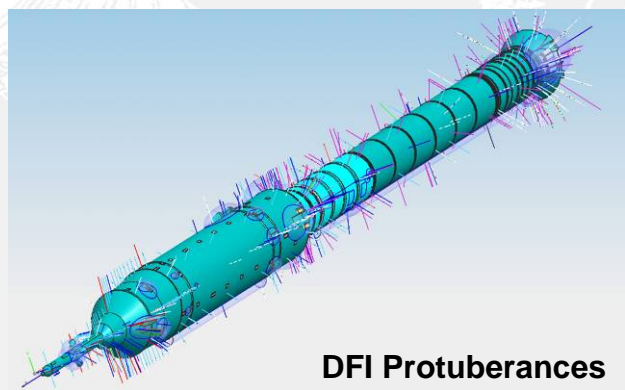


Element	Number of Sensors	% of Total
CMLAS	149	20%
SM, SA, US 1-7	156	21%
Interstage 1&2	62	8%
1st Stage	384	51%
Total	751	100%

Note: Totals include Pyro Sensor Counts



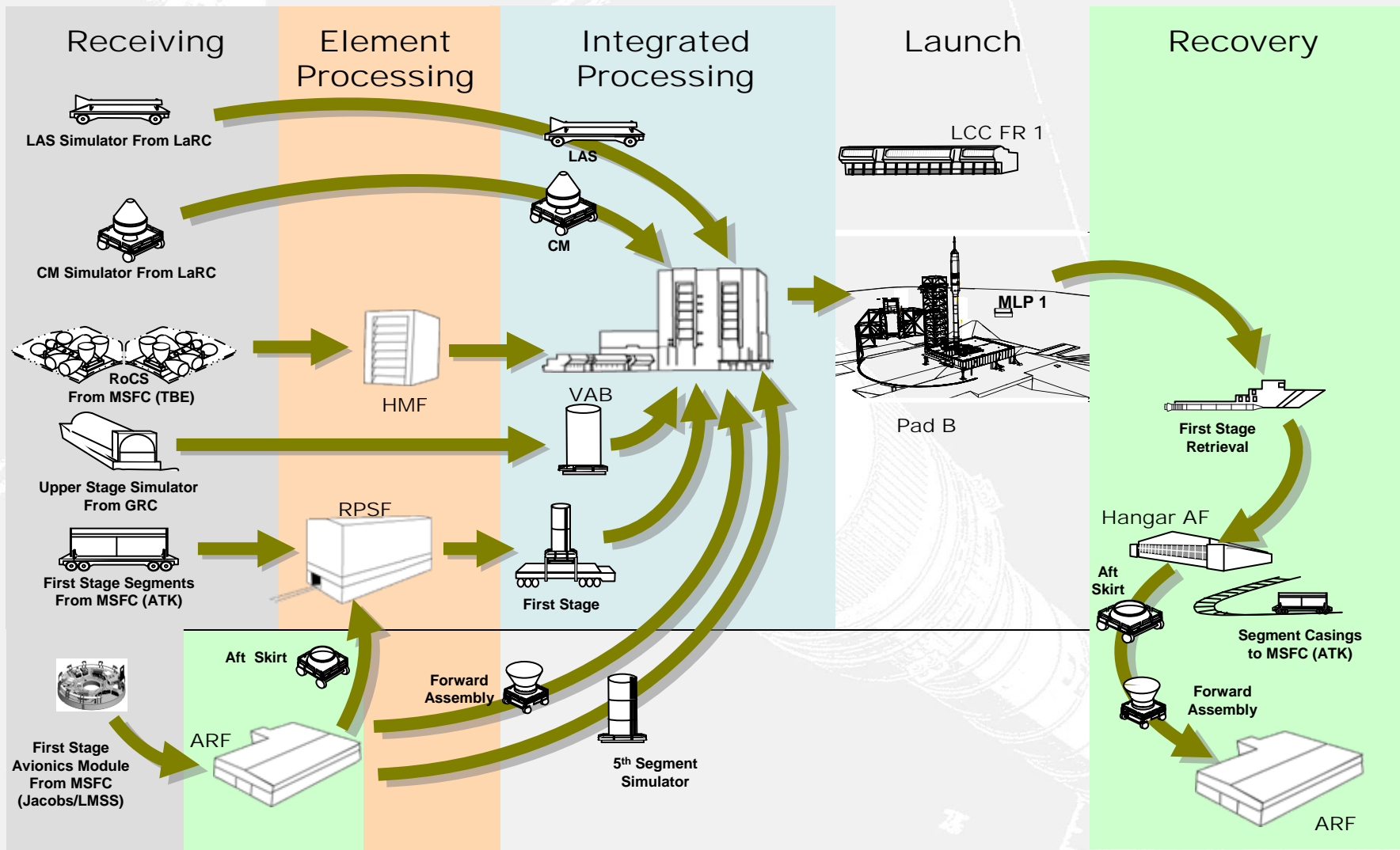
- ◆ Instrumented for 924 measurements
  - Thermal
  - Structures
  - GNC/Trajectory
  - Aero
  - Shock
- ◆ Cameras strategically located
- ◆ Data to be retrieved via telemetry and a data recorder box that is recovered from the First Stage after flight
- ◆ Managed at the NASA Langley Research Center, Hampton, VA



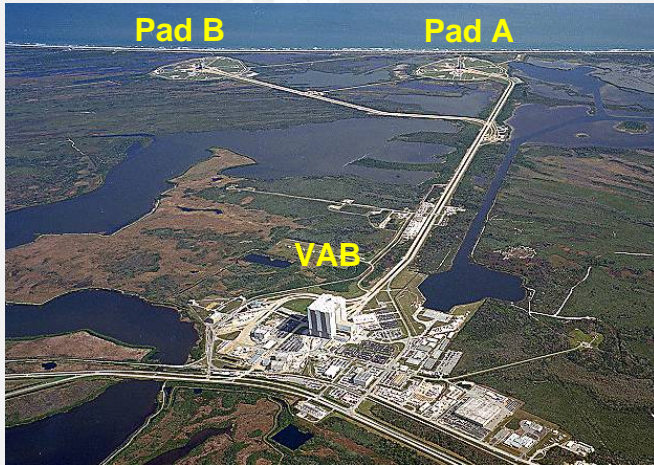
DFI Protuberances



# Ares I-X Processing Flow



# Vehicle Assembly Building (VAB) Operations



Aerial of the VAB, Pad B, and Pad A

- ◆ The Upper Stage Simulator (USS) segments and Orion Crew Module/Launch Abort System (CM/LAS) will be assembled into stacks and Development Flight Instrumentation (DFI) tested in VAB Hi-Bay 4.
- ◆ The First Stage segments and stacks will be integrated in Hi-Bay 3.



FTV in VAB



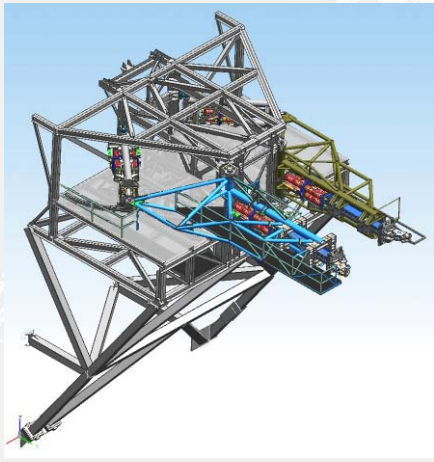
USS Segments and CM/LAS in Hi-Bay 4



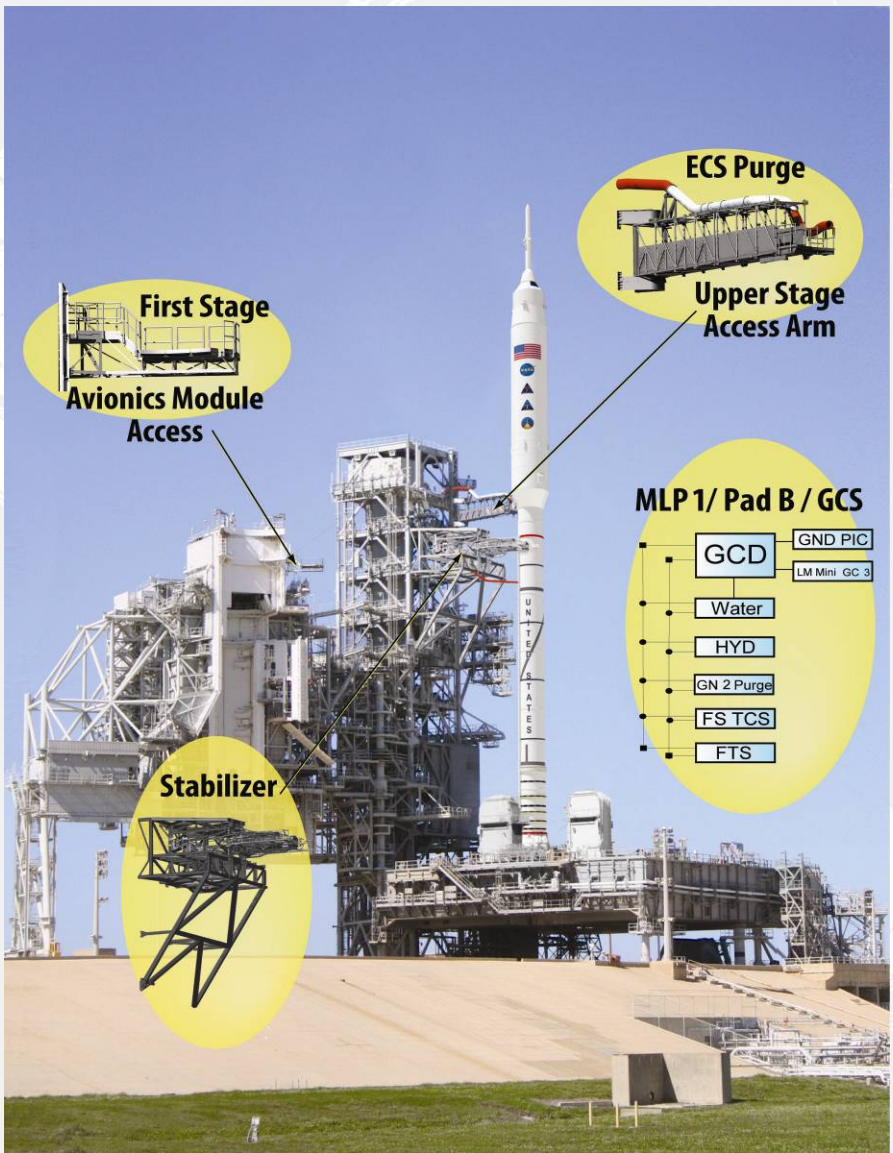
Ares I-X on Mobile Launch Platform



# Pad 39B at NASA Kennedy Space Center, FL



**Stabilizer**



**First Stage  
Avionics Module  
Access**

**ECS Purge  
Upper Stage  
Access Arm**

- MLP 1/ Pad B / GCS**
- GCD
    - GND PIC
    - LM Mini GC 3
  - Water
  - HYD
  - GN 2 Purge
  - FS TCS
  - FTS

**Stabilizer**



**Upper Stage Access Arm**

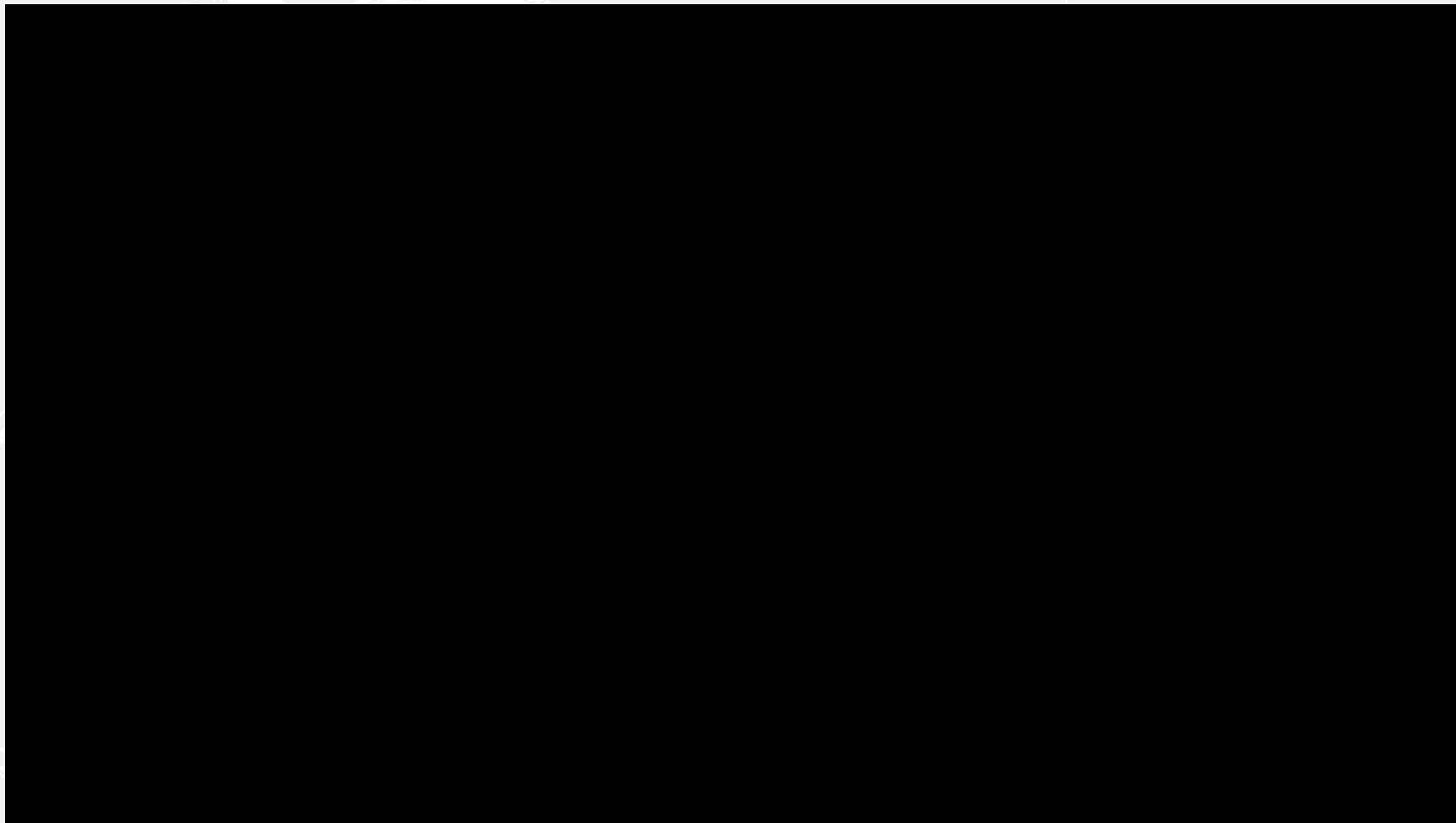


**1<sup>st</sup> Stage Avionics Module Access**

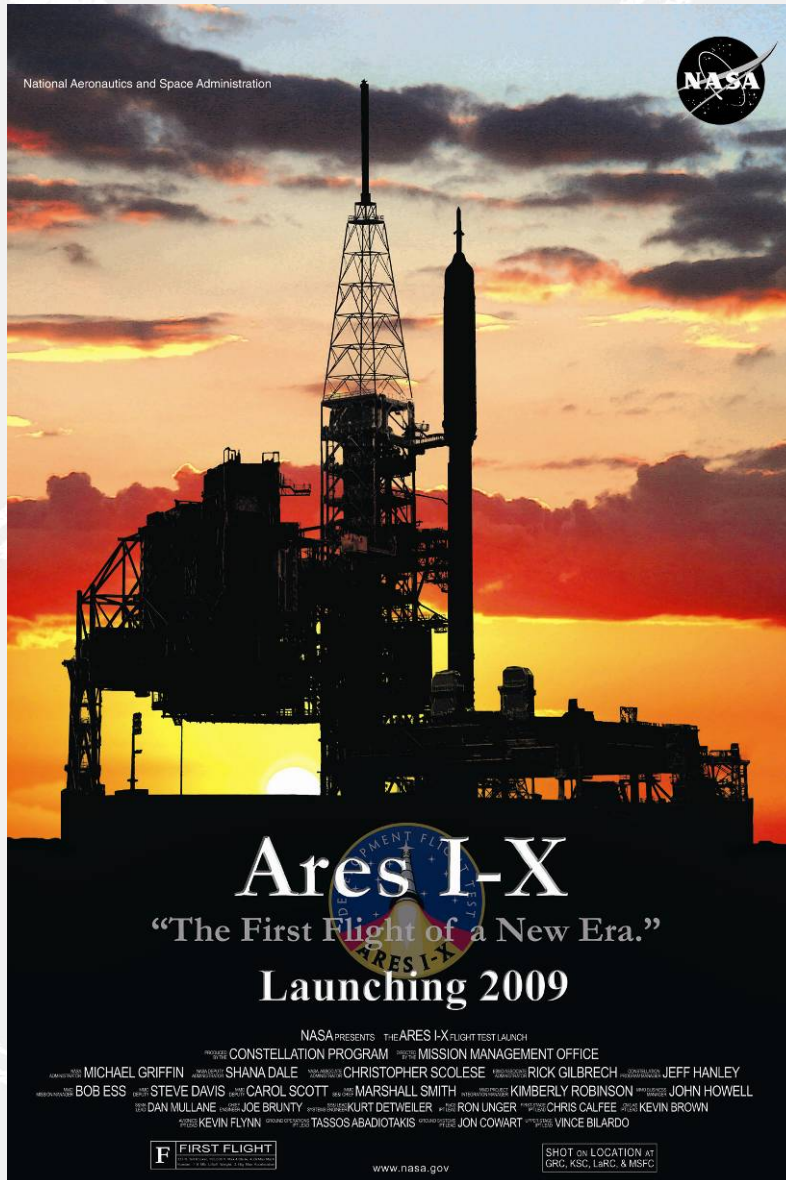


# Video

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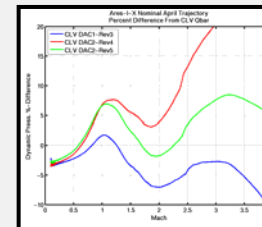


- ◆ Ares I-X is the first flight of NASA's new Constellation Program
- ◆ Ares I-X is a developmental test flight to support the Ares I
- ◆ Ares I-X is on track for May 2009 launch date
- ◆ For more information, see [http://www.nasa.gov/mission\\_pages/constellation/ares/flighttests/areslx/index.html](http://www.nasa.gov/mission_pages/constellation/ares/flighttests/areslx/index.html) or [http://staging.cms.nasa.gov/mission\\_pages/constellation/ares/flighttests/areslx/index.html](http://staging.cms.nasa.gov/mission_pages/constellation/ares/flighttests/areslx/index.html)





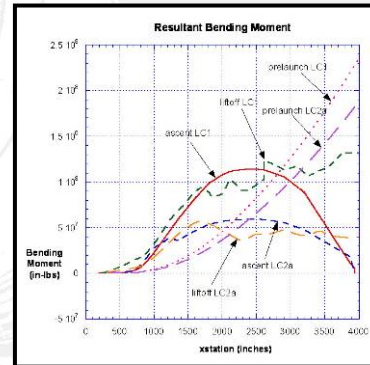
# Backup Slides



- Trajectories
- Malfunction Turn
- Baseline Databook
- Prelim Range Data Package

## Aerodynamics

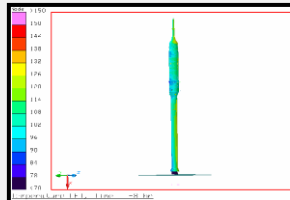
- Ascent Aero
- Transition Lift-off CFD
- Stage Separation CFD
- Rigid Buffet



## Structures

- Coupled Loads Cycle 1
- Cycle 2 Update
- Assess Rigid Buffet data

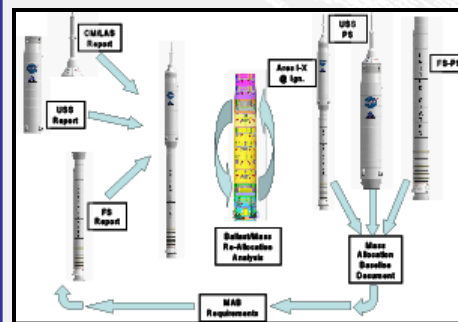
## Integrated Design & Analysis



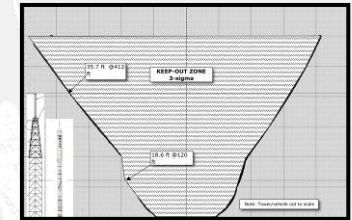
## Thermal

- Preliminary Stack
- Ascent on Pad
- Thermal Prediction Report & Databook

## Integrated Mass Properties



## -Mass Allocation



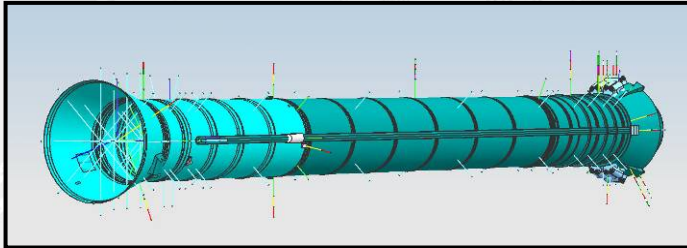
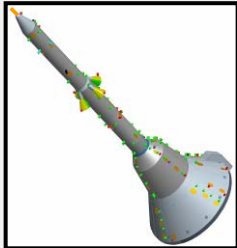
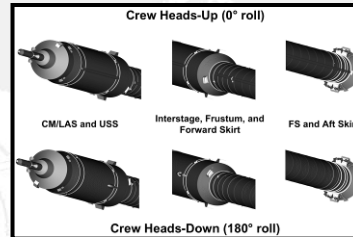
## Guidance, Navigation & Control

- Initial Flight Control Architecture
- Preliminary Stability Analysis
- Updated Drift Analysis
- Stage Separation Analysis

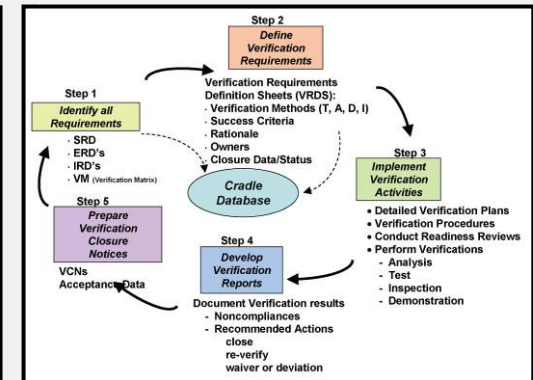
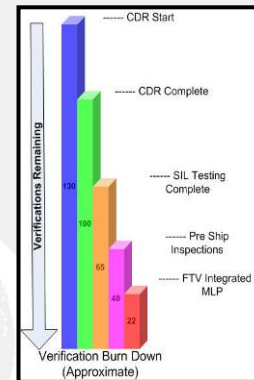
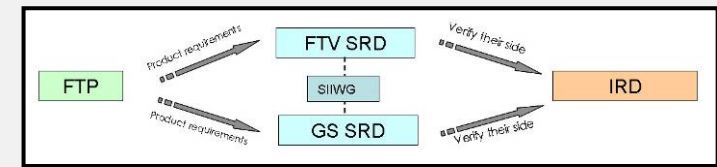


## Systems Engineering

- Outer Mold Line
- Development Flight Instrumentation
- Design Definition Document



- Systems Engineering
- System Requirements & Verification
- Launch Operations & Flight Integration



## System Requirements & Verification

- System Requirements Document
- Verification Requirements Document
- Interface Control Documents
- Interface Requirements Documents



## Launch Operations & Flight Integration

- Tailored Range Document
- Launch Commit Criteria
- Flight Data Package

# Ares I-X FTV Integration

